

Master

This publication has been superseded by the Airport

planning publication

APP

Publication (ACP)

Aircraft Characteristics BD500-3AB48-22000-00 Issue No. 032

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Publication No.: BD500-3AB48-22000-00

Manufacturer:



Airbus Canada Limited Partnership Customer Services 13100 Henri-Fabre Blvd., Mirabel, Quebec Canada J7N 3C6

Applicable to: All



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Applicable to: All

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Issue 032

The listed changes are introduced in Issue 032, dated 2023-10-19, of this publication.

Data module code Reason for change

BD500-A-J00-00-00-20AAA-018A-A Changed Data Module

To update the title to Aircraft Characteristics

This publication has been superseded by the BD500-A-J00-00-12AAA-030A-A Changed Data Module

To add A220 ACJ data

Aircraft Characteristics Publication_(ACP) BD500-A-J00-00-00-17AAA-030A-A Changed Data Module

To add A220 ACJ data

BD500-A-J00-00-00-11AAA-030A-A Changed Data Module

To add the A220 ACJ data

Applicable to: All



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Technical Publications Comment form

AIRBUS	A220
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TO: MCR FOCAL, TECHNICAL
PUBLICATIONS AIRBUS CANADA
LIMITED PARTNERSHIP

Name of airline:

13100, BOULEVARD HENRI-FABRE MIRABEL, QUEBEC, CANADA, J7N 3C6 E-MAIL ADDRESS: A220_UCFocal@abc.airbus All fields marked with an asterisk* are required		A220 reference #:	
		Date: dd-mmm-yyyy	
Contact information			
*Name:	*Corporation nar	me:	*Dept name/Code:
			he
Address:	City:		t by)- Province/State:
Postal code / Zip:	d	e ^{dCountry:}	(ACP*Telephone:
Mobile/Cell phone:		ers ti ^{O^{Fax nun}}	nber: n *E-mail:
n b ^{Publication} inform	request. NOTE: Respo	ke to receive notificationses will only be sent	by electronic mail
*Aircraft type:		e s P ^{U*Aircraft}	*Publication Module Code (PMC):
*Publication title/Issue:	as *Media Type: Paper Web	risti ^{C*Data Module C} (DMC):	*DMC issue date:
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Reason for change:	
Reference data provided: Yes No Description:	



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Applicable to: All

BD500-3AB48-22000-00



001 002

Change record

Check in the following record that all earlier changes has been incorporated.

Date by (signature)
Issue Incorporated

Issue Incorporated

Jul 29/2014 Initial issue 020-01 021

Date by (signature) on file Sep 19/2019

Dec 19/2014 BCSG Sep 05/2019 Signature Signature on file

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003 021-01 on file

Sep 15/2015 BCSG Oct 31/2019 Signature

Sep 24/2015 BCSG 021-02 Signature on file Nov 07/2019

Aircfaft Charaeteristics Publication (ACP)-

005 Mar 08/2016	BCSG 022	Nov 14/2019	Signature on file
006 007 008 009 010 011 012 Apr 20/2016 BCSG Nov 17/2016 BCSG May	May 18/2017 BCSC 15/2017 BCSG Oct May 12/2017 BCSG 022-01 023	024-01 025 Jun	Feb 06/2020 Signature on file Feb 20/2020 Signature on file Aug 27/2020 Signature on file Sep 17/2020 Signature on file le
013	Jan 25/2018 026 BCSG	01 Feb 25/2021 Signature on	file
014 014-01 015 015-01 016 016-01 017 017-01 018 018-01 019 Feb 15/2018 BCSG Jun 07/2018 BCSG Jun 14/2018 BCSG	Jul 26/2018 BCSG Aug 16/2018 BCSG Aug 23/2018 BCSG Sep 20/2018 BCSG Oct 04/2018 BCSG Oct	028-0 28-01 029-00 030-00 30-01 031-00 31-01 031-02 Mar 04/2021 Signature of file Mar 18/2021 Signature on file May 06/2021 Signature on file May 20/2021 Signature on file	Jun 10/2021 Signature on file Jun 17/2021 Signature on file May 19/2022 Signature on file Oct 20/2022 Signature on file Oct 20/2022 Signature on file Sep 21/2023 Signature on file Sep 28/2023 Signature on file
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List of effective data modules

of pages
Applicable to

The listed documents are included in Issue 032, dated A220-100 Aircraft Char 2023-10-19, of this publication. BD500-A-J00-00-20AAA-018A-A C 2023-09-15 3 50001-54999

C = Changed data module N = New data module

Document title Data module code Issue date No.

This publication has been superseded by the act eristics -

Introduction

Aircraft description - 50001-54999 BD500-A-J00-00-00-12AAA-030A-A C 2023-09-08 34 Technical data Aircraft Charaeteristics

Ground maneuvering - Technical data

Terminal servicing - Technical data

Operating conditions - Technical data

Pavement data - Techni cal data

Derivative aircraft - Tech nical data

Scaled drawings - Tech nical data

BD500-A-J00-00-00-13AAA-030A-A 2015-09-01 14

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2019-10-22 33 50001-54999

BD500-A-J00-00-00-18AAA-030A-A 2022-09-30 40

Publication (ACP)-50001-54999 BD500-A-J00-00-00-17AAA-030A-A C

2023-09-08 14 50001-54999

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Table of contents

The listed documents are included in Issue 032, dated 2023-10-19, of this publication.

Document title Data module code Issue date Applicable to

50001-54999

A220-100 Aircraft Characteristics - Introduction BD500-A-J00-00-00-20AAA-018A-A 2023-09-15

publication has been superseded by the

Aircraft performance - Technical data BD500-A-J00-00-00-13AAA-030A-A 2015-09-01 50001-54999 Ground

maneuvering - Technical data BD500-A-J00-00-00-19AAA-030A-A 2019-10-22 50001-54999 Aircraft

Characteristics Publication (ACP)

Terminal servicing - Technical data BD500-A-J00-00-00-18AAA-030A-A 2022-09-30 50001-54999 Operating conditions - Technical data BD500-A-J00-00-00-17AAA-030A-A 2023-09-08 50001-54999 Pavement data - Technical data BD500-A-J00-00-00-11AAA-030A-A 2023-09-08 50001-54999 Derivative aircraft - Technical data BD500-A-J00-00-00-22AAA-030A-A 2019-10-22 50001-54999 Scaled drawings - Technical data BD500-A-J00-00-00-21AAA-030A-A 2019-10-22 50001-54999

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BD500-3AB48-22000-00 **A220-100** Aircraft Characteristics - Introduction

Applicability: 50001-54999

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Description

1 Scope of the publication

The A220-100 Aircraft Characteristics, prepared by Airbus, contains general data on the airport facilities, ramp, and runway areas necessary to operate the Airbus commercial aircraft model BD-500-1A10 (A220-100).

Since operational practices vary among airlines, specific data should be coordinated with the user airlines prior to facility design. For additional information, please contact Airbus.

The content of this publication will change as options and aircraft changes occur. Make sure that you refer to the latest release of this publication.

If there is a difference between the data contained in this publication and that given by the local regulatory authority, the data from the local regulatory authority must be obeyed.

2 Publication organization

This publication is divided into eight sections:

- Aircraft description
- Aircraft performance
- Ground maneuvering

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See applicability on the first page of the DM BD500-A-J00-00-00-20AAA-018A-A BD500-A-J00-00-00-20AAA-018A-



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- Terminal servicing
- Operating conditions
- Pavement data
- Derivative aircraft
- Scaled drawings

3 Dimensions and weight

Linear dimensions given in this publication are in inches. The metric equivalents are given in parentheses ().

This publication has been superseded by the

Weight measures is given in pound (lb) with the metric equivalent in parentheses ().

4 Correspondence

Aircraft Characteristics Publication (ACP)

The publications change request form is available online and is used to request technical changes to rectify any errors, omissions, or procedural inconsistencies (if applicable), etc. using the Airbus A220 Interactive Electronic Technical Publication (IETP) viewer.

5 Translation of publication

If all or part of this publication is translated, the official version is the English language version by Airbus.

6 Standard term definitions

Maximum design Taxi Weight (MTW)

Maximum de sign Landing Weight (MLW)

Maximum de sign Take **Off Weight** (MTOW)

Operational **Weight Empty** (OWE)

Maximum design Zero **Fuel Weight** (MZFW)

Maximum car go volume

on the ground. This in cludes the fuel for these displacements and the takeoff run.

Maximum weight for landing as limited by aircraft strength and airworthiness requirement.

Maximum weight for take off as limited by aircraft strength and airworthiness requirements. This includes weight of fuel for taxi and run-up.

Weight of structure, power plant, furnishings, systems, unusable fuel and other items of equipment that are a necessary part of a particular aircraft configuration. Also included are certain standard items, personnel, equip

ment and supplies necessary for full operations, but does not include usable fuel or payload.

Maximum weight permitted before usable fuel and other usable agents must be loaded in defined sections of the aircraft, as limited by strength and air worthiness requirements.

The maximum space available for cargo.

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BD500-A-J00-00-00-20AAA-018A-A



BD500-3AB48-22000-00

Maximum seating capac ity
The maximum number of passengers

permitted based on certification re quirements.

Usable fuel Fuel available for aircraft propulsion and the Auxiliary Power Unit (APU).

7 Acronyms

The first time an acronym is used it will be defined, and all subsequent uses will be in blue. When you mouse over the acronym the definition will appear. Acronyms are not plural in this publication.

This publication has been superseded by the Aircraft

Characteristics Publication (ACP)

See applicability on the first page of the DM BD500-A-J00-00-00-20AAA-018A-A

BD500-A-J00-00-00-20AAA-018A-A End of data module

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BD500-3AB48-22000-00 Aircraft description - Technical data

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Characteristics Publication (ACP)· References

Data Module/Technical Publication Title

None

Description

1 Aircraft characteristics

1.1 Introduction

This data module contains general data about the Airbus model BD-500-1A10 (A220-100) char acteristics. The structural weight limits, such as maximum ramp weight, and zero fuel weight are dependent on configuration. Refer to each aircraft's specified Weight and Balance Manual (WBM) BD500-3AB48-22100-00 and weight and balance report for structural limits and other weight information.

Refer to Table 2 and Table 3 for the aircraft characteristics.

Refer to Table 4 for the system fluid capacities.

Refer to Table 5 for the service fluid capacities.

1.2 Aircraft characteristics

Applicability: 50001-50061, 50063-50065, 50068, 50070, 50072-50077, 50079-54999

Table 2 Aircraft characteristics

Description	A220-100	
Engines	2 Pure Power™ PW1519G ¹	
Mode	Passenger	

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-

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BD500-3AB48-22000-00

Description	A220-100
Standard seating capacity	120
Maximum Ramp Weight (MRW)	141,500 lb (64,183 kg)
Maximum Take-Off Weight (MTOW)	140,500 lb (63,730 kg)
Maximum Landing Weight (MLW)	120,500 lb (54,658 kg)
Maximum Zero Fuel Weight (MZFW)	116,000 lb (52,617 kg)
Minimum Flight Weight (MFW)	77,000 lb (34,927 kg)
Maximum fuel tank capacity	
	the

	5,756 US gal (21 805 L)	
Unusable fuel		
	by) · 220.5 lb (100 kg)	
Maximum cargo volume - Overhead bins d CP ^{280 ft³ (7,93 m³)}		
e d (A ¹ Optional engine models: PW1521G and PW1524G		

lication

Supers^e Applicability: 50062, 50066-50067, 50069, 50071, 50077-50078, 50081

Table 3 Aircraft characteristics

Table 3 Aircraft characteristics			
e ^{nDescription}	UDA220-100		
b ^e Engines	P _{2 Pure Power™ PW1519G} 1		
s ha ist ^{j_{Mode}}	Passenger		
n te ^r Maximum Ramp Weight (MRW)	135,000 lb (64,183 kg)		
o ati ra ^C Maximum Take-Off Weight (MTOW)	134,000 lb (60,781 kg)		
C Ii ha Maximum Landing Weight (MLW)	112,500 lb (51,029 kg)		
pu ft C ^{Maximum} Zero Fuel Weight (MZFW)	108,000 lb (48,987 kg)		
is ra Minimum Flight Weight (MFW)	77,000 lb (34,927 kg)		
h T ir ^C 1 Optional engine models: PW1521G and PW1524G			

1.3 System fluid capacities

Description	Volume	Weight	
Engine fluids calculated with 8.24 lb/US gal (0,987 kg/L)			
Engine oil tank at 60 °F	6.5 US gal (24.4 L)	53.1 lb (24.1 kg)	
Lines and internal engine oil	7.7 US gal (29.2 L)	63.5 lb (28.8 kg)	
APU fluids calculated with 7.98 lb/US gal (0.956 kg/L)			

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-

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Description	Volume	Weight
APU oil tank	1.94 US gal (7.3 L)	15.4 lb (7.0 kg)
APU lines and internal oil	0.84 US gal (3.2 L)	6.7 lb (3.0 kg)
Hydraulic fluids at 77 °F (25 °C) low density 8.20 lb/US gal (0.983 kg/L)		
System No. 1 reservoir	5.0 US gal (18.8 L)	40.8 lb (18.5 kg)
System No. 2 reservoir	4.33 US gal (16.4 L)	35.50 lb (16.1 kg)
System No. 3 reservoir	4.33 US gal (16.4 L)	35.50 lb (16.1 kg)
Systems and lines	34.6 US gal (131.0 L)	
		th^e 283.8 lb (128.7 kg)

ed by

1.4 Service fluid capacities

Table 5 Service fluid capacities

CP)·

Description	Volume	d e (^{Aweight}
S er tio ^{NPotable} water at 60 °F (15,5 °C)		
Galley/Lavatory tank level 100%	UP 41.9 US gal (158.8 L)	ca ^{350.0 lb (158.8 kg)}
S N UDI ^I Chemical toilet fluid at 60 °F (15,5 °C)		

Waste tank level 100%	be P ^{38 US gal (143.8}	317.1 lb (143.8 kg)
-----------------------	----------------------------------	---------------------

Aircraft Characteristic⁹

This publication has **2 Aircraft dimensions**

2.1 General aircraft dimensions

This section contains general data about the aircraft dimensions.

72 ft 9 in. (22,2 m)

64 ft 5 in. (19,6 m)

111 ft 9 in. (34,1 m)

32 ft 11 in. (10,0 m)

BD500-3AB48-22000-00

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-

A 2023-09-08 Page $_{(4,9 \text{ m})}^{16 \text{ ft } 1.5 \text{ in.}}$

8 ft 2 in.



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11.5 ft.

Aircraft Cha <u>ra</u> cteris	tics Publication (<u>AC</u>P)·
115	i ft 1 in. (35,1 m)
40	ft 3 in. (12,3 m)
	22 ft 1 in. (6,7 m)
	2 ft 3 in. (0,7 m)
	BD500-A-J00-00-00-12AAA-030A-A
See applicability on the first page of the DM ICN-BD500-A-J000000-A-3AB48-22469-A-002-01 Figure 1 General aircraft dimensions - (Sheet 1 of 2)	BD500-A-J00-00-00-12AAA-030A-A 2023-09-08 Page 5
BD500-3AB48-22000-	⁰⁰ This publication has <u>been</u>
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-	

12 ft. 2.5 in. (3,7 m)

11 ft. 7 in<u>.</u> (3,5 m)

Aircraft

Characteristics

114 ft. 9 in. (34,9 m)

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43 ft. (13,1 m)

ICN-BD500-A-J000000-A-3AB48-22470-A-003-01 Figure 1 General aircraft dimensions - (Sheet 2 of 2)

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See applicability on the first page of the DM



BD500-3AB48-22000-00 2.2 General aircraft area

Table 6 General aircraft area

Description	A220-100
ESDU wing area (including ailerons, flaps, spoilers and area within the fuselage)	1209 ft² (112.3 m²)
Total horizontal stabilizer area (hor izontal tail area and elevator area)	395 ft² (36.6 m²)
Total vertical stabilizer area (ver tical tail area and rudder area)	304 ft² (28.2 m²)
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3 Ground clearances

This section gives the height of various points of the aircraft, above the ground.

Dimensions in the tables are approximate and will vary with tire type, weight and balance and other special conditions.

3.1 Ground clearances

first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-

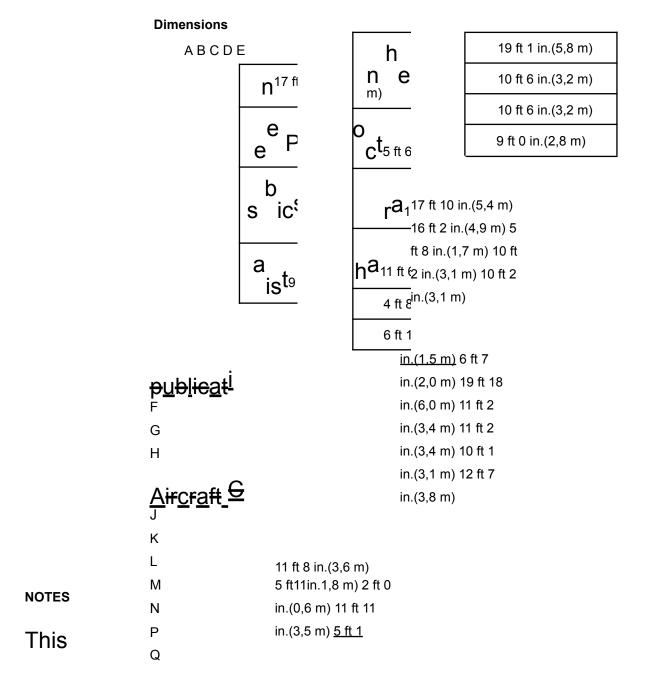
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See applicability on the



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Vertical clearances shown are the greatest possible variations in attitude due to the variation of aircraft weight and center of gravity.

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Figure 2 Ground clearances

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See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A

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Table 7 Ground clearances for evacuation slides

Description	Dimensions
Forward Passenger Door (FPD) Slide	240 in. (6096 mm)
Forward Service Door (FSD) Slide	240 in. (6096 mm)
Aft Passenger Door (APD) Slide	200 in. (5080 mm)
Aft Service Door (ASD) Slide	
	e 200 in. (5080 mm)
Overwing Emergency Exit Door (OWEED) Slides (Left & Right sides)	h by t)· 119 in. (3022.60 mm)

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first page of the DM BD500-A-J00-00-00-12AAA-030A-A

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200.00 in.

Aircraft Characteriscs blication-(ACP) · (3022.60 mm) 240.00 in. (6096.00 mm) t ESCAPE 1 119.00 in. (3022.60 mm) 200.00 in. (5080.00 mm) 1 SLIDE ESCAPE SLIDE

NOTE

1

See applicability on the first page of the DM Figure 3 Ground clearances for evacuation slides

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1 Emergency evacuation ground area.

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BD500-3AB48-22000-00 4 Layout of passenger

compartment accommodation

The passenger compartment includes the galley area, lavatory, and passenger seating area.

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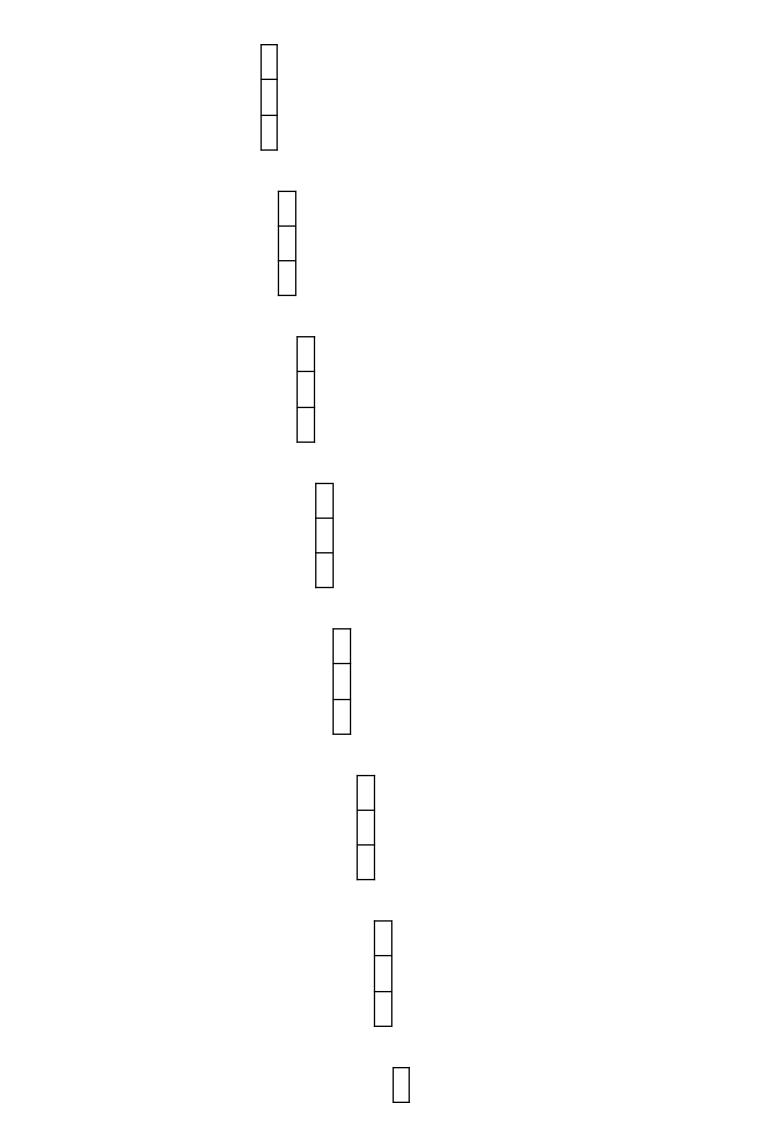
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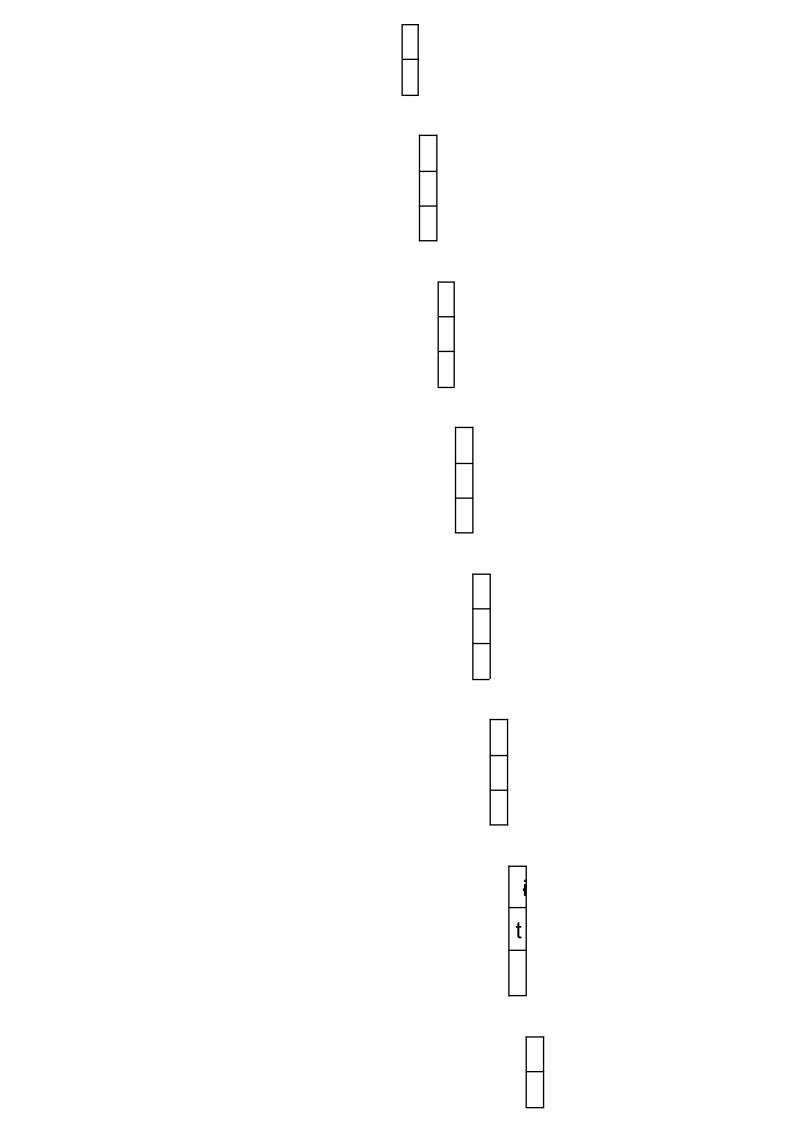
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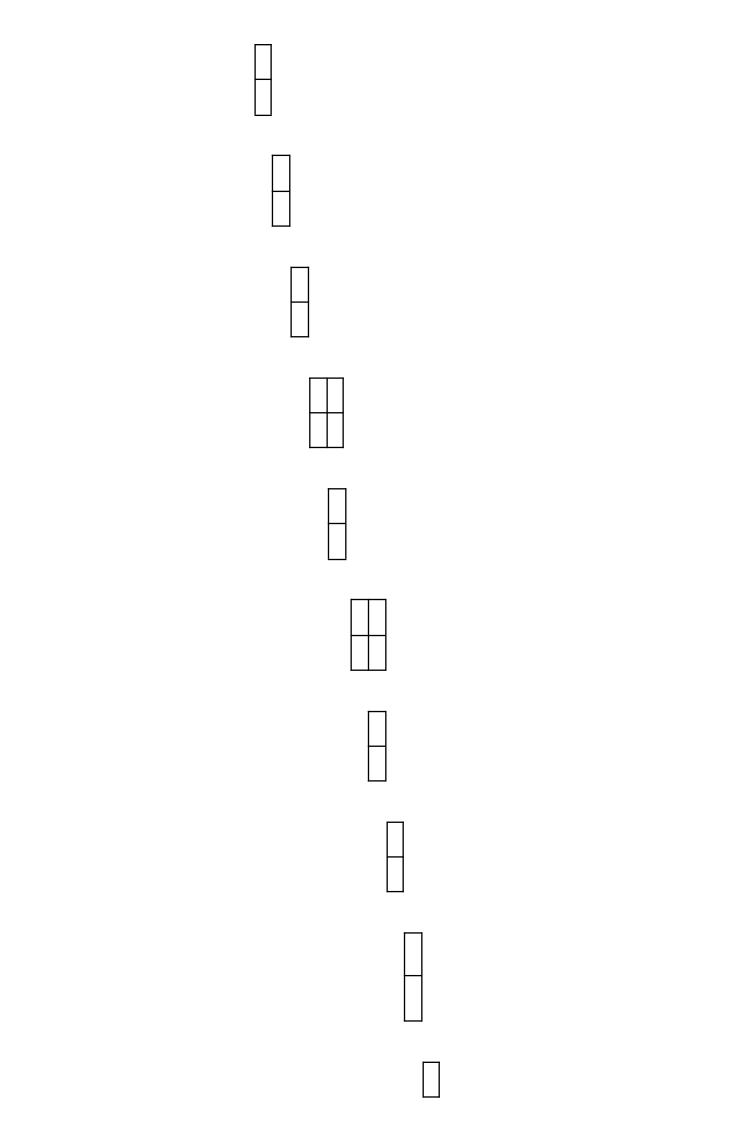
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en (ACP)





Aircraft Characteri s tics P ubl i <u>e</u> ^a	



32 in. (81,3 cm)-42 in. (106,7 cm)

14 x 32 in. (81,3 cm) ⁶ x 33 in. (83,8 cm)

ICN-BD500-A-J061200-A-3AB48-00007-A-001-01

Figure 4 Layout Of Passenger Accommodation (LOPA)

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BD500-3AB48-22000-00 5 Passenger cross-section

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Characteristics Publication (ACP)

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18.5 in. (0,47 m) TYPICAL 4 PLACES

129.0 in. (3,28 m) 19.0 in. (0,48 m)

29.4 in.

by the (0,75 m)

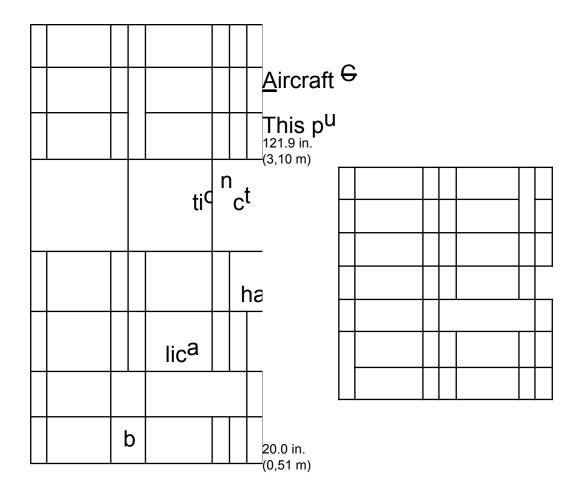
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(ACP):

84.0 in. (2,13 m)

as been supers ded





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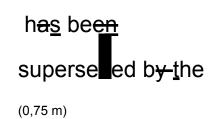
See applicability on the first page of the DM ICN-BD500-A-J061200-A-3AB48-00010-A-001-01 Figure 5 Passenger cross-section (economy class)

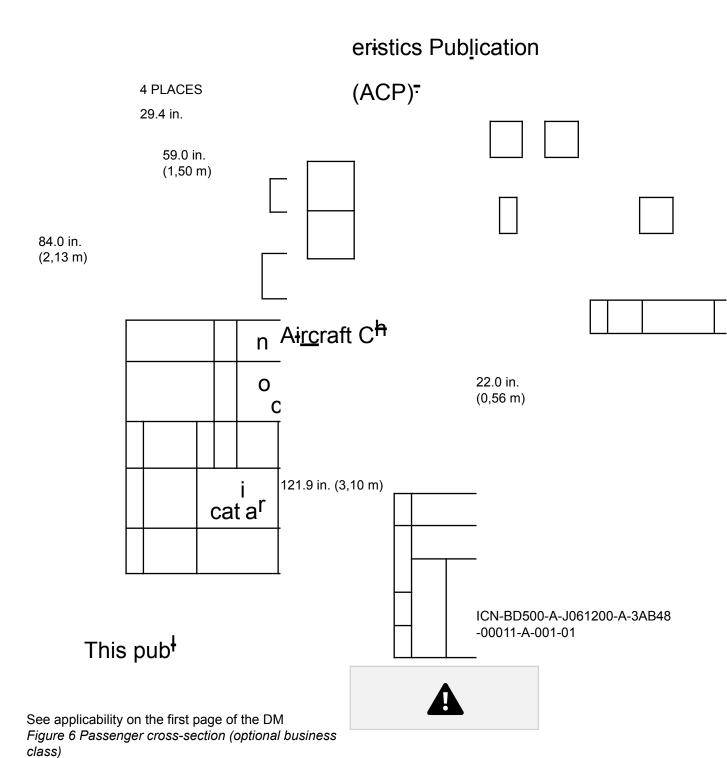
129.0 in. (3,28 m) 20.0 in. (0,51 m) TYPICAL

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2023-09-08 Page 15

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11 x 17 x 25 in.
(64 x 43 x 28 cm)

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ASSOCIATION (IATA) ROLLER BAG

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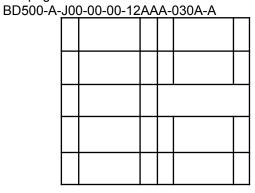
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6 Cargo compartment BD500-3AB48-22000-00

ICN-BD500-A-J061200-A-3AB48-00012-A-001-01 Figure 7 Overhead stowage bins

BD500-A-J00-00-00-12AAA-030A-A

ment and the Environmental Control System (ECS) distribution bay. The aft compartment is po sitioned between the mid equipment compartment and the water system bay. Refer to Fig. 8.

Both compartments are furnished with heavy duty floor panels and sidewall linings and are sealed to meet the requirements of a Class C compartment. Decompression and ventilation panels are provided as well. The compartment linings also incorporate provisions for compart ment lighting, smoke detector, and fire extinguish.

The combined maximum weight loading of the cargo compartment is 8,290 lb (3 760 kg).

6.1 Cargo door nets

uperseded by the
Protective nets are provided at the door area of each cargo compartment to prevent

baggage ication (ACP).

from fouling the door due to in-flight shifting of the loads. Refer to Fig. 9.

6.2 Volumes - Cargo compartment

The estimated volume of the cargo compartments is based on geometric volume and accounts for the unusable area in the vicinity of the cargo doors. Table 8 lists the estimated wet volume of the cargo compartments.

Table 8 Cargo compartment volumes

Description	n	S _U Usable Volume	bl _{Maximum load}	
	ft ³	e p ^{m3}	lb	kg
Fwd cargo compartment	ha ³¹	S	3742	1697
		risti ^{C_{8.80}}		
Aft cargo com partment	n tio a ⁴²²	Cte _{11.95}	4548	2063

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192.9 in. (4.90 m)

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266.48 in. (6.77 m)

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BD500-A-J00-00-00-12AAA-030A-A 2023-09-08 Page 19

See applicability on the first page of the DM

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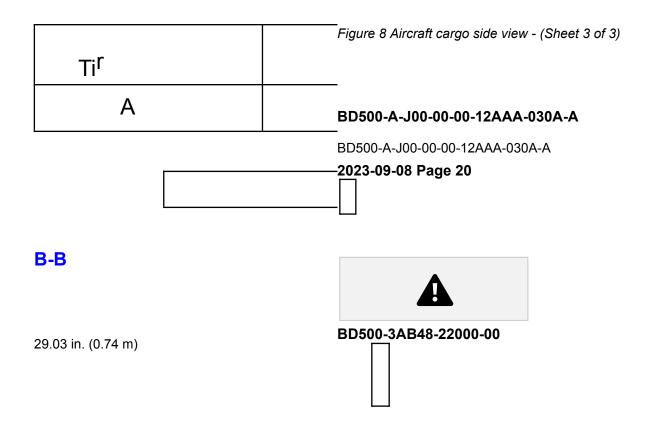
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See applicability on the first page of the DM ICN-BD500-A-J084305-A-3AB48-10440-A-001-01 This publication has been superseded by the Aircraft Characteristic Publication (ACP)

> ICN-BD500-A-J502200-C-3AB48-17798-A-001-01 Figure 9 Cargo nets - (Sheet 1 of 2)

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-

A 2023-09-08 Page

CARGO COMPARTME NT NET BD500-3AB48-22000-00



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NET CUP NET CUP NET CUP		

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A Figure 9 Cargo nets - (Sheet 2 of 2)

BD500-A-J00-00-00-12AAA-030A-A

2023-09-08 Page 22



BD500-3AB48-22000-00 7 Door clearances and clear

opening dimensions A general description of the doors is as follows:

7.1 Passenger/Crew

Two semi-plug type doors on the left side of the aircraft provide access for passengers and crew. Door 1L is considered the primary entrance while door 2L provides a secondary entrance available for passenger loading/unloading as well as ground servicing.

Each door is classified as a type C floor level exit. Due to the sill height, every door incorporates an emergency evacuation slide system. In addition each one translates outwards from closed position, supported by a hinged arm to rest in open position.

This publication has been superseded by the

Every door is operable from the exterior and interior of the aircraft and features an inspection window to allow verification of the outside conditions from the interior. The exterior operating handle has a linear motion and is interconnected to a vent flap system to provide pressure

Aircraft Characteristics Publication (ACP)

equalization between the aircraft and the ambient air prior to be opened.

Each door is fully lined and insulated to meet thermal and noise performance requirements.

For Passenger/Crew doors distance from the nose, refer to Fig. 11Fig. 12 . For aft passenger door opening and clearances, refer to Fig. 13 .

7.2 Emergency exit

The over-wing emergency exits are type III semi-plug type doors.

The exits are provided with an operating handle with removable cover and are fitted with a stan dard sized passenger compartment window. Each door is fully lined and insulated to meet ther mal and noise performance requirements.

The door rotates upwards from the closed position, supported by a hinged arm to rest in open position. The door opening sequence is automatically supported by the energy stored in its own mechanism.

For emergency access to the passenger compartment, the doors may be opened from an exterior handle.

Due to the exit path height from the ground, an off-wing evacuation slide system is provided.

For over-wing emergency exits distance from the nose, refer to Fig. 11 . For doors dimensions, refer to Table 9 .

7.3 Flight compartment emergency exit

The flight compartment is outfitted with a single, inward-opening overhead escape hatch.

7.4 Cargo doors

Access doors are provided to allow cargo compartment loading and unloading.

The semi-plug forward and aft cargo doors are identical components, each hinged along the top edge of its frame.

Each door incorporates an exterior lock/unlock handle with linear motion that is interconnected to a vent flap system and provide pressure equalization between the aircraft and the ambient air prior to be opened.

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See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A

BD500-A-J00-00-00-12AAA-030A-



BD500-3AB48-22000-00 Each door is fully lined and insulated to meet thermal and noise performance requirements.

For cargo doors distance from the nose, refer to Fig. 11 . For doors dimensions, refer to Table 9 . For forward cargo door opening and clearances, refer to Fig. 14 . For aft cargo door opening and clearances, refer to Fig. 15 .

7.5 Service doors

Two semi-plug type doors are provided on the right side of the aircraft to provide access for the forward (door 1R) and aft (door 2R) galley service areas.

Each door is classified as a type C floor level exit. Due to the sill height, each door incorporates an emergency evacuation slide system.

This publication has been superseded by the

Each door translates outwards from the closed position, supported by a hinged arm and stabiliz ing system, to rest parallel to the fuselage in the open position.

Each door is operable from the exterior and interior of the aircraft and features an

inspection Aircraft Characteristics Publication (ACP)

window to allow verification of the outside conditions from the interior. The exterior operating handle has a linear motion and is interconnected to a vent flap system to provide pressure equalization between the aircraft and the ambient air prior to be opened.

Each door is fully lined and insulated to meet thermal and noise performance requirements.

For service doors distance from the nose, refer to Fig. 11 . For service doors dimensions, refer to Table 9 . For forward service door opening and clearances, refer to Fig. 16 . For aft service door opening and clearances, refer to Fig. 17 .

7.6 Forward avionics bay door

A plug-type door is provided in the forward fuselage to gain access to the pressurized forward equipment compartment. The door is fitted with a stowable operating handle.

For forward equipment compartment door distance from the nose, refer to Fig. 11 . For dimen sions, refer to Table 9 .

7.7 Mid avionics bay door

A plug-type door is provided in the mid fuselage to gain access to the pressurized mid equip ment compartment. The door is fitted with a stowable operating handle.

For mid equipment compartment door distance from the nose, refer to Fig. 11. For dimensions, refer to Table 9.

7.8 Aft equipment bay door

A door is provided in the aft fuselage to gain access to the unpressurized aft equipment compartment.

For aft equipment compartment door distance from the nose, refer to Fig. 11 . For dimensions, refer to Table 9 .

7.9 Doors identification

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See applicability on the first page of the DM



FORWARD PASSENGER DOOR

OVERWING EMERGENCY EXIT DOOR BD500-3AB48-22000-00

AFT PASSENGER DOOR

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FORWARD

AFT SERVICE DOOR

AFT COMPARTMENT CARGO DOOR

FORWARD

first page of the DM BD500-A-J00-00-00-12AAA-030A-A ICN-BD500-A-J000000-A-3AB48-23216-A-001-01 Figure 10 General door location

BD500-A-J00-00-00-12AAA-030A-A

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See applicability on the



BD500-3AB48-22000-00 7.10 Access and exit doors

dimensions

Door	Height	Width
Main entrance door - Type C exit (door 1L)	6 ft 3 in. (1,9 m)	2 ft 6 in. (0,8 m)
Service door - Type C exit (door 1R)	5 ft 0 in. (1,5 m)	2 ft 6 in. (0,8 m)
Aft entrance door - Type C exit (door 2L)	6 ft 0 in. (1,8 m)	2 ft 6 in. (0,8 m)
Service door - Type C exit (door 2R)	5 ft 0 in. (1,5 m)	2 ft 6 in. (0,8 m)
Forward avionics bay door	2 ft 8 in. (0,81 m)	
		he 3 ft 8 in. (1,1 m)
Mid avionics bay door	2 ft 8 in. (0,81 m)	t
		y). 3 ft 8 in. (1,1 m)

Aft equipment bay door	d b _{3 ft 6 in. (1,08 m)}	CP ^{1 ft 11 in. (0,6}
Forward cargo compartment door	ed ^{e2} ft 8 in. (0,81	(A ^{3 ft 8 in. (1,1 m)}
Aft cargo compartment door	S r io ^{n2 ft 8 in.} (0,81 m)	3 ft 8 in. (1,1 m)
Over-wing emergency exit	e up ca ^{t_{3 ft 6 in.}}	1 ft 11 in. (0,59 m)
Flight compartment emergency exit	S bl ⁱ 22 in. (0,559 m)	20 in. (0,508 m)

This publication has be Aircraft Characteristics Pth 7.11 Door distance from nose

first page of the DM BD500-A-J00-00-12AAA-030A-A BD500-A-J00-00-12AAA-030A-

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<u>Gharacteristics</u> Publication (ACP)

D E G K

Dimensions A220-100

A	L
В	
С	NOTE 102 ft 4 in. (31.2 m) 93 ft 10 in. (28.6
D	m) 84 ft 4 in. (25.7 m) 69 ft 4 in. (21.1
E	m) 58 ft 0 in. (17.7 m) 43 ft 0 in.
F	(13.11 m) 21 ft 4 in. (6.5 m) 15 ft 10
G	in. (4.8 m) 15 ft 0 in. (4.6 m) 14 ft 3
Н	in. (4.3 m) 9 ft 2 in. (2.8 m)
J	() = (= ,
К	

The values shown are the greatest possible variations in attitude due to the variation of aircraft weight and gravity.

ICN-BD500-A-J000000-A-3AB48-21712-A-003-01

Figure 11 Door distance from nose

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BD500-3AB48-22000-00 7.12 Door opening and clearance

This publication has been superseded by the Aircraft

Characteristics Publication (ACP)

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-

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superseded by the



24.39 in. (61,95 cm) MINIMUM

73.85 in. (178,58 cm) MINIMUM

VIEW LOOKING DOWN

ICN-BD500-A-J061100-A-3AB48-00103-A-003-01 Figure 12 Forward passenger door opening and clearances

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A

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23.88 in. (60,65 cm) MINIMUM



30.78 in. (78,18 cm) MINIMUM

73.85 in. (187,58 cm)

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ICN-BD500-A-J061100-A-3AB48-00104-A-003-01 Figure 13 Aft passenger door opening and clearances

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-



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33 in. (83,82 cm) MINIMUM

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Figure 14 Forward cargo compartment door opening and clearances

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33 in. (83,82 cm) MINIMUM

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Figure 15 Aft cargo compartment door opening and clearances

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See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-



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23.12 in. (58,72 cm) MINIMUM 24.92 in. (63,30 cm) MINIMUM

> 72.71 in. (184,68 cm) MINIMUM

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Figure 16 Forward service door opening and clearance

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See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A BD500-A-J00-00-00-12AAA-030A-

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Aircraft Characteristics Publication (ACP) 30.77 in. (78,16 cm)

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23.85 in. (60,59 cm) MINIMUM

74.63 in. (189,57 cm) MINIMUM

BD500-A-J00-00-00-12AAA-030A-A End of data module

See applicability on the first page of the DM BD500-A-J00-00-00-12AAA-030A-A

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BD500-3AB48-22000-00 Aircraft performance - Technical data

Applicability: 50001-54999

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Aircraft performance - Technical data	
References Description	
Introduction	
Payload/Range	² Thi s
<u>publication</u> has been superseded by t	
Landing field length requirements	
reference speed	¹³ Aircr a ft
Characteristics Publication (ACP)	List of tables Page
1 Deferences	
1 ReferencesStandard day temperature chart	
	List of figures Page
1 Zero Fuel Weight (ZFW) vs Range ISAfield length - ISA - PW1519Glength ISA +15°C - PW1519G	5 3 Takeoff field
ISA - PW1521G	
+15°C - PW1521G	
PW1524G	

References

Table 1 References

Data Module/Technical Publication Title

None

Description

1 Introduction

This data module gives data about:

- Payload/Range
- Takeoff field length requirements
- Landing field length requirements
- Landing reference speed

The table below provides standard day temperature for pressure altitudes.

A 2015-09-01 Page

See applicability on the first page of the DM BD500-A-J00-00-00-13AAA-030A-A

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Table 2 Standard day temperature chart

BD500-3AB48-22000-00

Altitude		Standard day temperature	
Feet (ft.)	Meters (m)	٥F	°C
0	0	59	15
2000	610	51.9	11
4000	1220	44.7	7.1
6000	1830	37.6	3.1
8000	2440	30.5	
			he -0.8
10000	3050	23.3	y)4.8

This publication has been superseded Aircraft Characteristics Publication (ACP 2 Payload/Range

This section gives information about the payload/range at ISA conditions.

first page of the DM BD500-A-J00-00-00-13AAA-030A-A

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See applicability on the



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Aircraft Characteristics Publication (ACP)

00-A-J00 01 0000-A-3 AB48-238 Ib) (1000 ZFW ICN-BD5

BD500-A-J00-00-00-13AAA-030A-A

2015-09-01 Page 3

See applicability on the first page of the DM Figure 1 Zero Fuel Weight (ZFW) vs Range ISA



BD500-3AB48-22000-00 3 Takeoff field length

requirements

For more information about aircraft performance, refer to the Aircraft Flight Manual (AFM) BD500-3AB48-22200-00.

For aircraft performance and field length requirements refer to:

- Fig. 2 for the takeoff field length ISA PW1519G.
- Fig. 3 for the takeoff field length ISA +15°C PW1519G.
- Fig. 4 for the takeoff field length ISA PW1521G.
- Fig. 5 for the takeoff field length ISA +15°C PW1521G.
- Fig. 6 for the takeoff field length ISA PW1524G.

This publication has been superseded by the - Fig. 7 for the takeoff field length ISA +15°C - PW1524G.

Aircraft Characteristics Publication (ACP)

first page of the DM BD500-A-J00-00-00-13AAA-030A-A BD500-A-J00-00-00-13AAA-030A-

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ICN-BD500-A-J000000-A-3AB48-01753-A-002-01 Figure 2 Takeoff field length - ISA - PW1519G

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Page 5



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See applicability on the first page of the DM Figure 3 Takeoff field length ISA +15°C - PW1519G BD500-3AB4 (8-22000-00

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Aircraft Characteristics
Publication (ACP)

-29045-A-001-01 Figure 4 Takeoff field length ISA - PW1521G

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BD500-A-J00-00-00-13AAA-030A-A

2015-09-01 Page 8

See applicability on the first page of the DM Figure 5 Takeoff field length ISA +15°C - PW1521G

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BD500-3A B48-22000

Publication (ACP)

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-29047-A-001-01 Figure 6 Takeoff field length ISA - PW1524G

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See applicability on the first page of the DM BD500-A-J00-00-00-13AAA-030A-A

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Aircraft Characteristics Publication (ACP):

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BD500-A-J00-00-00-13AAA-030A-A

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See applicability on the first page of the DM Figure 7 Takeoff field length ISA +15°C - PW1524G

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BD500-3AB48-22000-00 4 Landing field length

requirements

For more information about landing field, refer to the AFM BD500-3AB48-22200-00. For landing field length requirements refer to Fig. 8 .

This publication has been superseded by the Aircraft

Characteristics Publication (ACP)

See applicability on the first page of the DM BD500-A-J00-00-00-13AAA-030A-A BD500-A-J00-00-00-13AAA-030A-

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Characteristics Publication (ACP)

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Landing Field Lenght (1000 ft)

BD500-A-J00-00-00-13AAA-030A-A

See applicability on the first page of the DM ICN-BD500-A-J000000-A-3AB48-01757-A-002-01 Figure 8 Landing field length - Dry runway

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BD500-3AB48-22000-00 5 Landing reference speed

This section gives information about the landing reference speed.

This publication has been superseded by the Aircraft

Characteristics Publication (ACP)

See applicability on the first page of the DM BD500-A-J00-00-00-13AAA-030A-A BD500-A-J00-00-00-13AAA-030A-

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Characteristics Publication (ACP)

kt) (Vref Speed

Reference Landing

BD500-A-J00-00-00-13AAA-030A-A ICN-BD500-A-J000000-A-3AB48-23901-A-002-01 Figure 9 Landing reference speed



BD500-3AB48-22000-00 Ground maneuvering - Technical data

Applicability: 50001-54999

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Ground maneuvering - Technical data				
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Introduction				
Turning radii				
Aireraft Characteristics Publication (ACP)				
Ground maneuvering - Technical data				
A220-100 turning radii for various nose wheel angles				
List of figures Page				
Visibility from cockpit in static position				
1 References				
None				

Description

1 Turning radii

1.1 Introduction

This data module contains data about the aircraft turning capability and maneuvering character istics on the ground. The data is based on aircraft performance in good conditions of operation.

A 2019-10-22 Page

See applicability on the first page of the DM BD500-A-J00-00-00-19AAA-030A-A

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BD500-A-J00-00-00-19AAA-030A-



BD500-3AB48-22000-00

Thus, the values must be considered theoretical and used only as an aid. Refer to Table 2 for the values to use with Fig. 1 for the turn radii with 3 degree slip angle.

1.2 Landing gear turning radii, including minimum turning radiiThis

publication has been superseded by the

Aircraft Characteristics Publication (ACP)

BD500-A-J00-00-00-19AAA-030A-

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See applicability on the



BD500-3AB48-22000-00 Table 2 A220-100 turning radii for

various nose wheel angles					
Turning angle (in degrees) with 3 degree tire slip	Turning center to air craft center line (D)	Nose tip (R1)	Nose gear outside face (R2)	Mai	
17	1686.8 in.	1807.5 in.	1776.3 in.		
	(42844.72 mm)	(45910.50 mm)	(45118.02 mm)		
27	1012.1 in.	1202.7 in.	1148.4 in.		
	(25707.34 mm)	(30548.58 mm)	(29169.36 mm)		
37	684.4 in. (17383.76 mm)	943.6 in (23967.44 mm)	869.3 in. (22080.22 mm)		
47	480.9 in.	808.3 in	717.6 in.		
	(12214.86 mm)	(20530.82 mm)	(18227.04 mm)		
57	334.9 in.	730.9 in.	627.9 in.		
	(8506.46 mm)	(18564.86 mm)	(15948.66 mm)	SI	
67	218.9 in.	685.5 in.	572.7 in.		
	(5560.06 mm)	(17411.70 mm)	(14546.58 mm)	be	
77	119.1 in. (3025.14 mm)	660.4 in. (16774.16 mm)	ha ^{S^{541.7 in} (13759.18 mm)}	ris	

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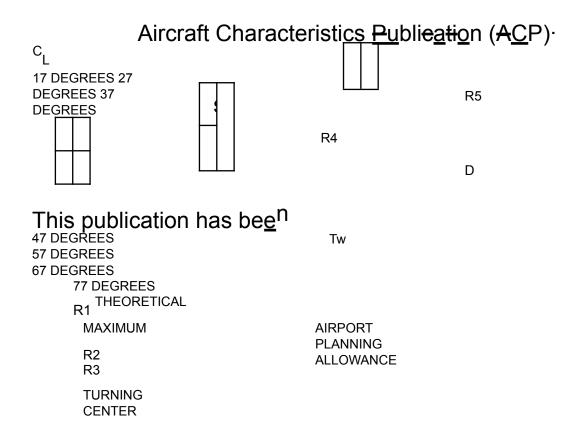
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5.0 ft (1,52 m)

first page of the DM BD500-A-J00-00-00-19AAA-030A-A ICN-BD500-A-J092001-A-3AB48-00068-A-001-01 Figure 1 Turn radii

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See applicability on the



BD500-3AB48-22000-00 2 Visibility from cockpit in

static position

This section contains data about the visibility from cockpit in static position. To see the diagram, refer to Fig. 2 .

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6



25°

17°

13 ft 3 in (4 m.) DOWN VISION (2,6 m.) BD500-3AB48-22000-00

8 ft 6 in

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43 ft 3 in (0,8 (13,1 m.) 2 ft 8 in

Aircraft Characteristics Publication (ACP). VISUAL ANGLES IN VERTICAL PLANE THROUGH PILOT'S EYE POSITION

106°

20 in. (0,5 m)

123°

VISUAL ANGLES IN HORIZONTAL PLANE THROUGH PILOT'S EYE POSITION

20 in. (0,5 m)

25°

NOTES
VISUAL ANGLE IN A PLANE
PERPEDNDICULAR TO LONGITUDINAL
AXIS THROUGH PILOT'S EYE POSITION

1. Not to be used for landing approach visibility. 2. Not scale.

Figure 2 Visibility from cockpit in static position

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2.1 Clear areas of vision

To see the diagram, refer to Fig. 3 and Fig. 4.

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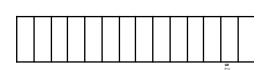
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LEFT RIGHT AZIMUTH ANGLE - DEGREES

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Figure 3 Clear areas of vision

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See applicability on the first page of the DM BD500-A-J00-00-00-19AAA-030A-A

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Figure 4 A220 Clear areas of vision

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BD500-3AB48-22000-00 3 Runways and taxiways turn

paths

This section contains data about the runways and taxiways turn paths.

3.1 More than 90 $^{\circ}$ turn - Runway to taxiway - Cockpit over centerline method To see the diagram, refer to Fig. 5 .

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RUNWAY CENTERLINE

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Aircraft Characteristics Publication (ACP) FAA LEAD-IN FILLET L = 150 ft.(45 m)

APPROXIMATE 26 ft. (8 m)

FILLET R =55 ft. (16.5 m)

> TAXIWAY CENTERLINE

> > 50 ft (15 m)

100 ft. (30 m)

LEGEND

Nose gear. Main gear.

NOTE

Coordinate with airline operator for the specific planned operating procedure.

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3.2 More than 90° turn - Runway to taxiway - Oversteering method To see

the diagram, refer to Fig. 6.

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Aircraft Characteristics Publication (ACP)

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RUNWAY CENTERLINE

TURN R = 100 ft. (30 m)

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Characteristics Publication (ACP)

FAA LEAD-IN FILLET L = 150 ft. (45 m)

APPROXIMATE 10 ft. (3.2 m)

FILLET R = 55 ft. (16.5 m)

TAXIWAY CENTERLINE

50 ft. (15 m)

100 ft. (30 m)

LEGEND

Nose gear. Main gear.

NOTE

Coordinate with airline operator for the specific planned operating procedure.

ICN-BD500-A-J000000-A-3AB48-22553-A-001-01
Figure 6 More than 90° turn - Runway to taxiway - Oversteering method

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3.3 90° turn - Runway to taxiway - Cockpit over centerline method To see

the diagram, refer to Fig. 7.

Aircraft Characteristics Publication (ACP) first page of the DM

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See applicability on the



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RUNWAY CENTERLINE

FAA LEAD-IN FILLET

TURN

Aircraft Characteristics

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supersede d by the R = 100 ft. (30 m) M

Main gear.

L = 150 ft. (45 m)₅₀ ft. (15 m)

NOTE

TAXIWAY CENTERLINE

APPROXIMATE 26 ft. (8.0 m)

FILLET

R = 55 ft. (16.5 m)

LEGEND

100 ft. (30 m)

Nose gear.

Coordinate with airline operator for the specific planned operating procedure.

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Figure 7 90° turn - Runway to taxiway - Cockpit over centerline method

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